

# Faster Simulations of the National Airspace System

PK Menon  
Monish Tandale  
Sandy Wiraatmadja  
*Optimal Synthesis Inc.*

Joseph Rios  
*NASA Ames Research Center*



NVIDIA GPU Technology Conference 2010, San Jose, CA

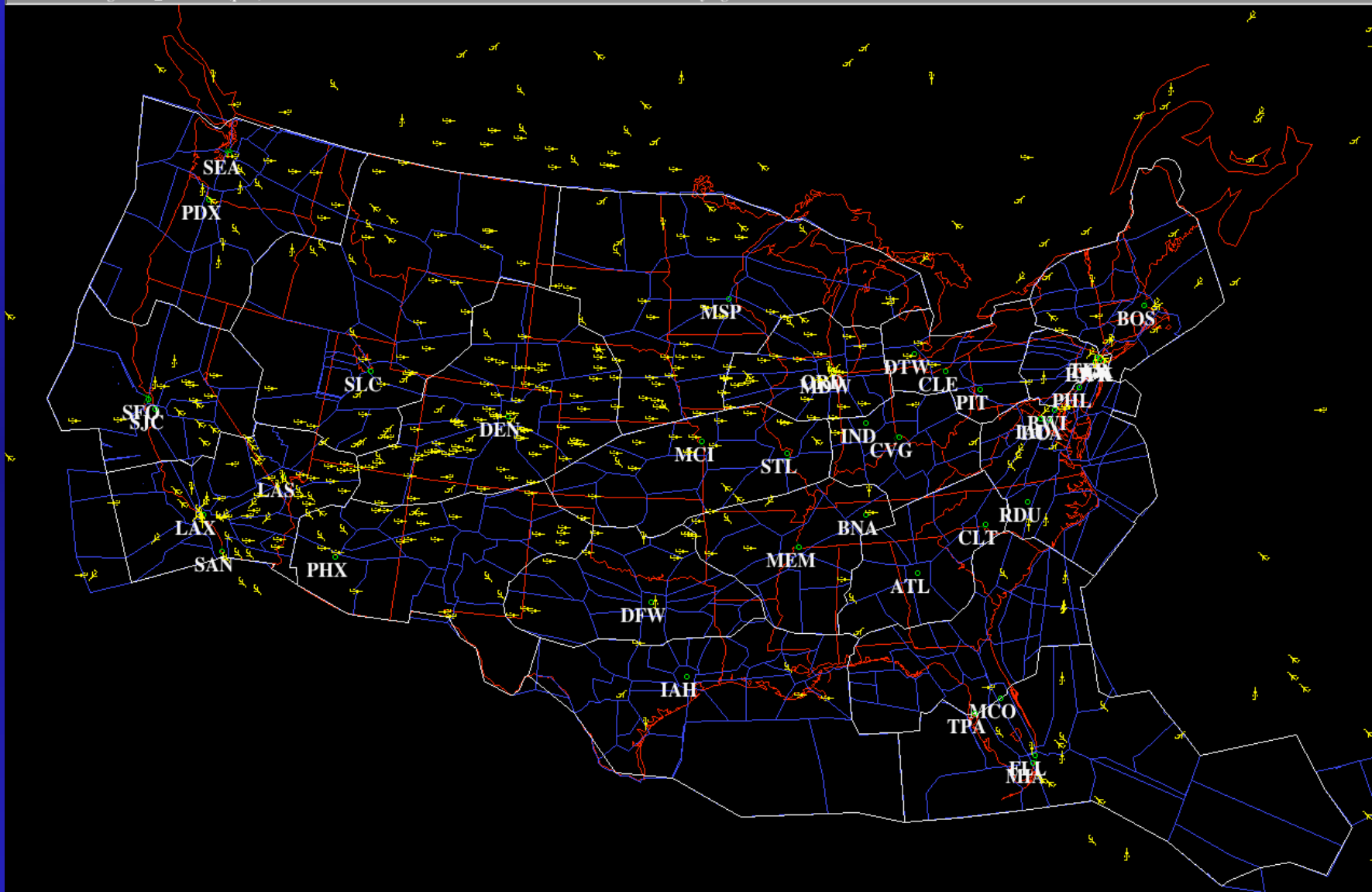


Status: Running ASDI\_07172006 (pbk)

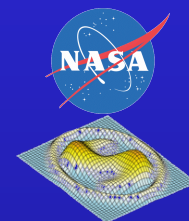
Number Flying: 587

NASA  
Ames Research Center

7/17/2006 08:00:13 UTC



7/17/2006 04:00 EDT

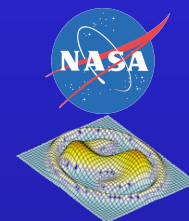


# Research Achievements

---

- Parallelized trajectory prediction code of a high-fidelity airspace simulator
- Ported parallel implementation to CUDA architecture
- Gained 240X speed-up over original implementation

*24-hour simulation with 35,000 aircraft  
completed under 2.5 seconds*

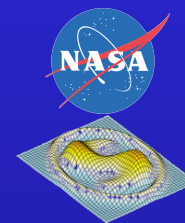


# Outline

---

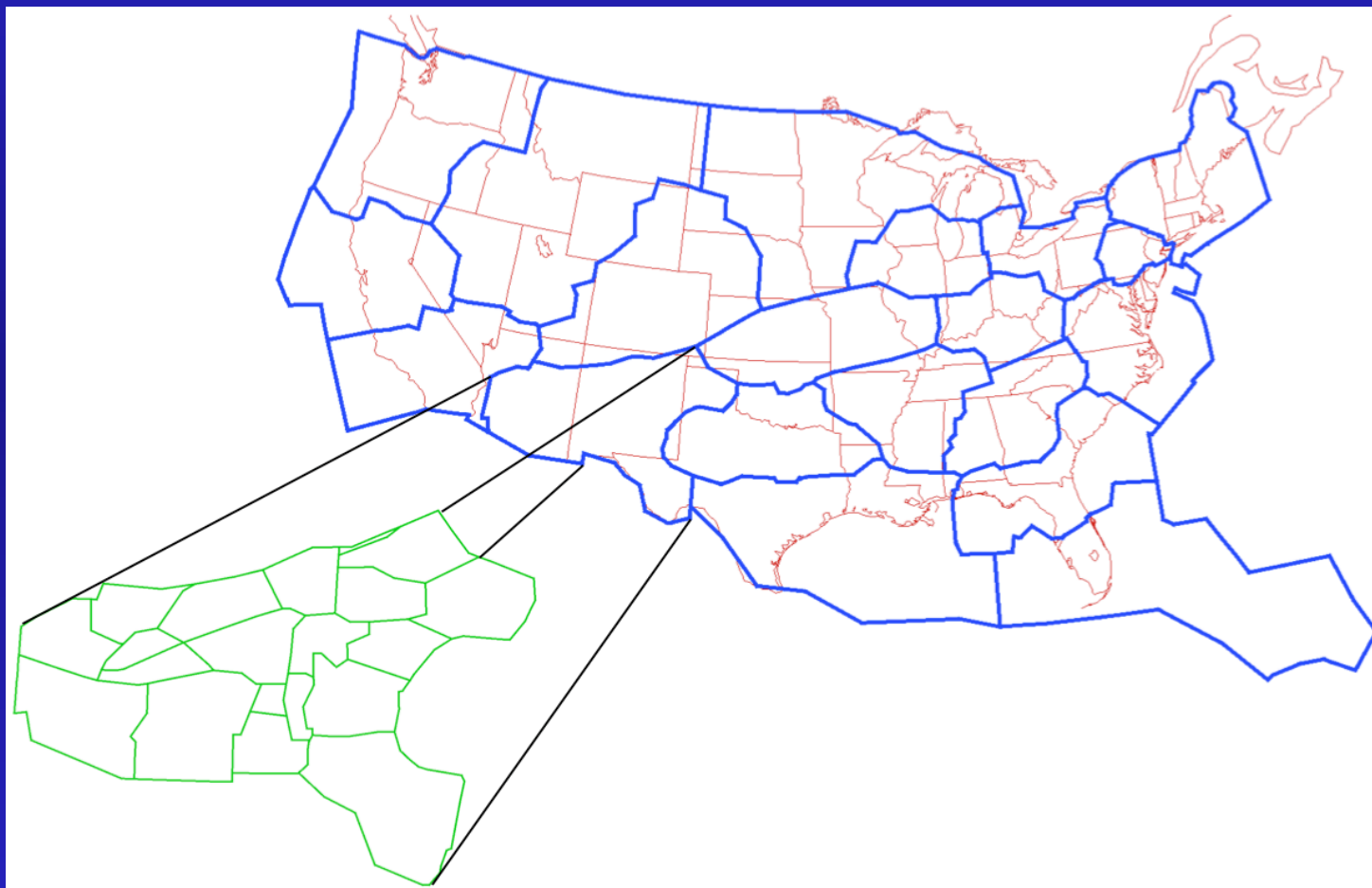
- Background
  - Air Traffic Management overview
  - Airspace Simulations
  - Traffic Flow Management
- Accelerating simulations for Traffic Flow Management Research
- Porting code to Compute Unified Device Architecture
- Experimental results
- Concluding remarks

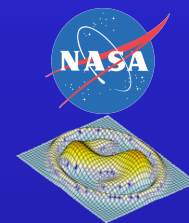
**BACKGROUND**



# Air Traffic Management

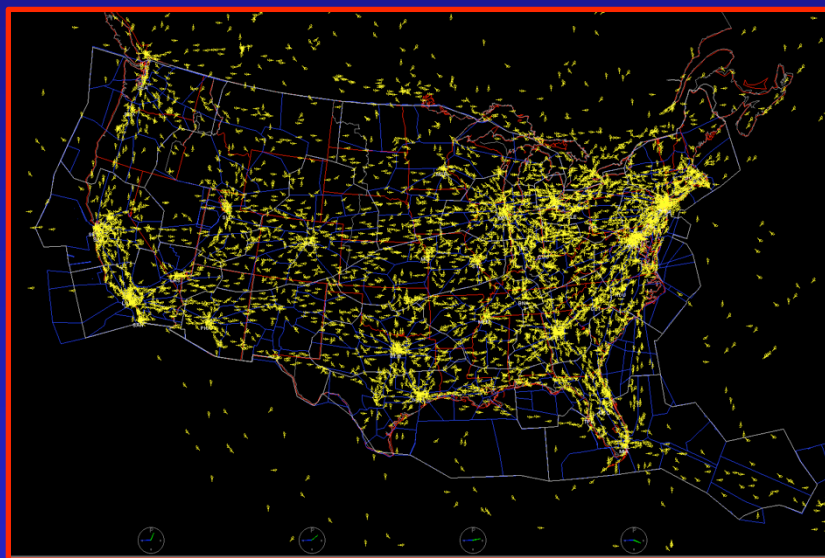
---

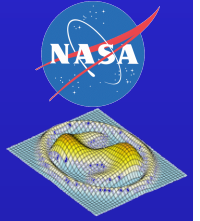




# Simulating Air Traffic

- FACET: Future ATM (Air Traffic Management) Concepts Evaluation Tool
- Java front end, C back end
- Takes traffic and weather data as input
- Works in one of two modes: Simulation or Playback
- In Simulation Mode, position of each aircraft propagated forward through calculations at each time step
- Through GUI or an API, user can query or modify state of simulation





# Traffic Flow Management Problem Definition

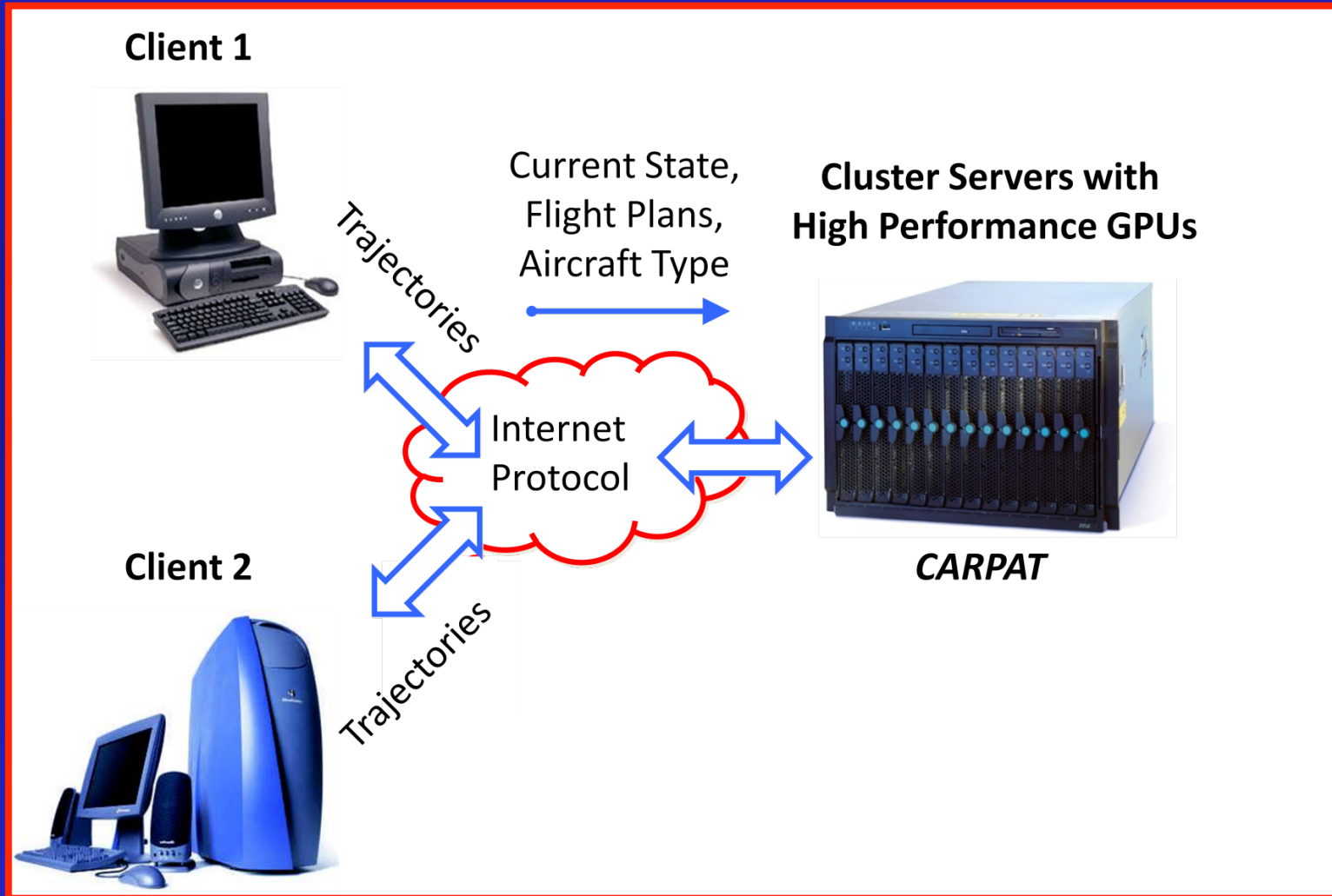
---

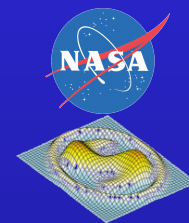
- Given a set of scheduled flights and a set of capacity values, how should those flights be held to minimize delay costs while respecting capacities?
- Search for solutions involves simulation of air traffic to build schedules and check capacity violations
- Future ATM Concepts Evaluation Tool (FACET) is powerful, commonly-used simulator



**ACCELERATING  
SIMULATIONS FOR  
TRAFFIC FLOW  
MANAGEMENT RESEARCH**

# Computational Appliance for Rapid Prediction of Aircraft Trajectories (CARPAT)

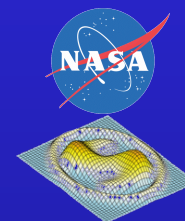




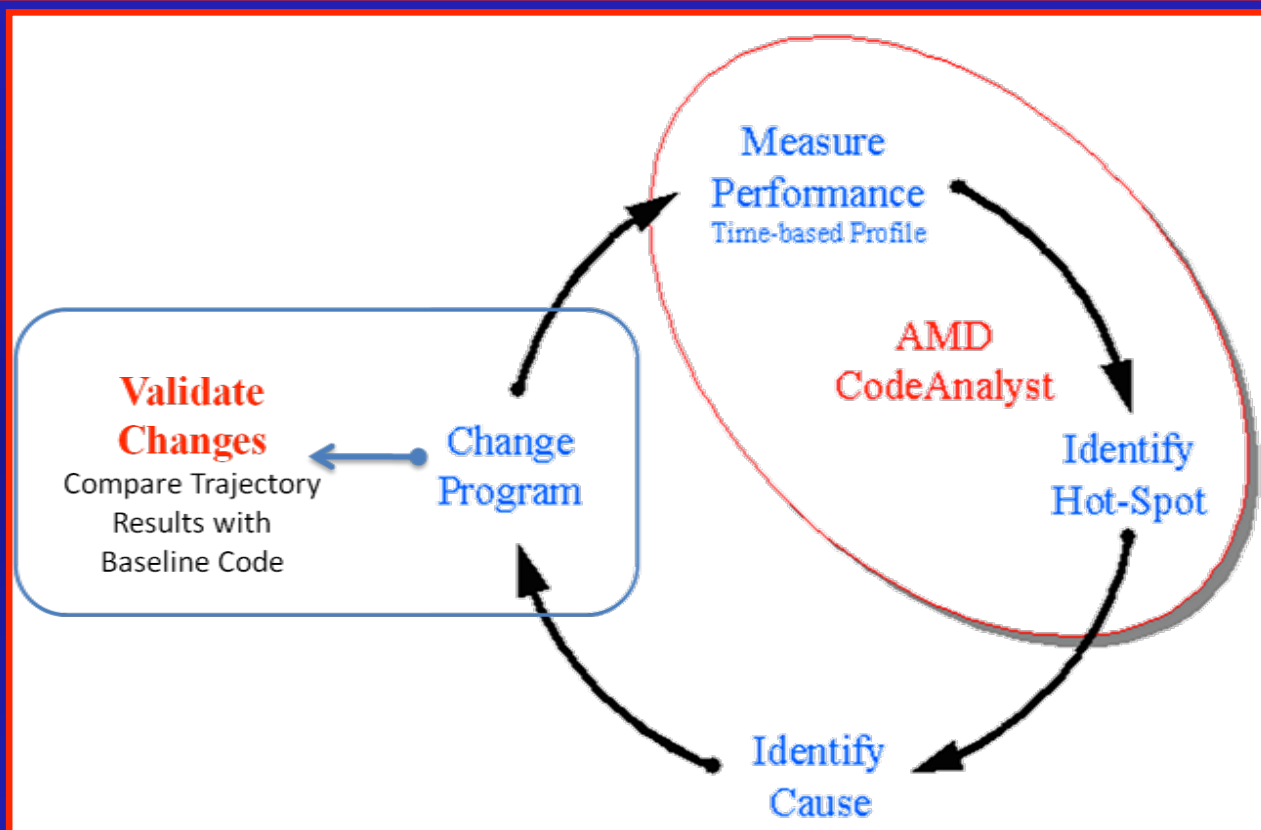
# Accelerating Simulations of Traffic Flow

---

- Software Profiling and Tuning
- Parallelization
  - Cluster Implementation
  - Multi-Core Implementation
  - GPU Implementation
    - GPU as a co-processor
    - GPU as the primary processor

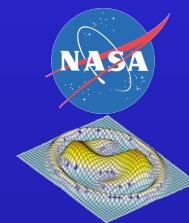


# Software Profiling and Tuning



- Optimized table lookup, optimized linked list management, elimination of redundant calculations in case of no change of state.

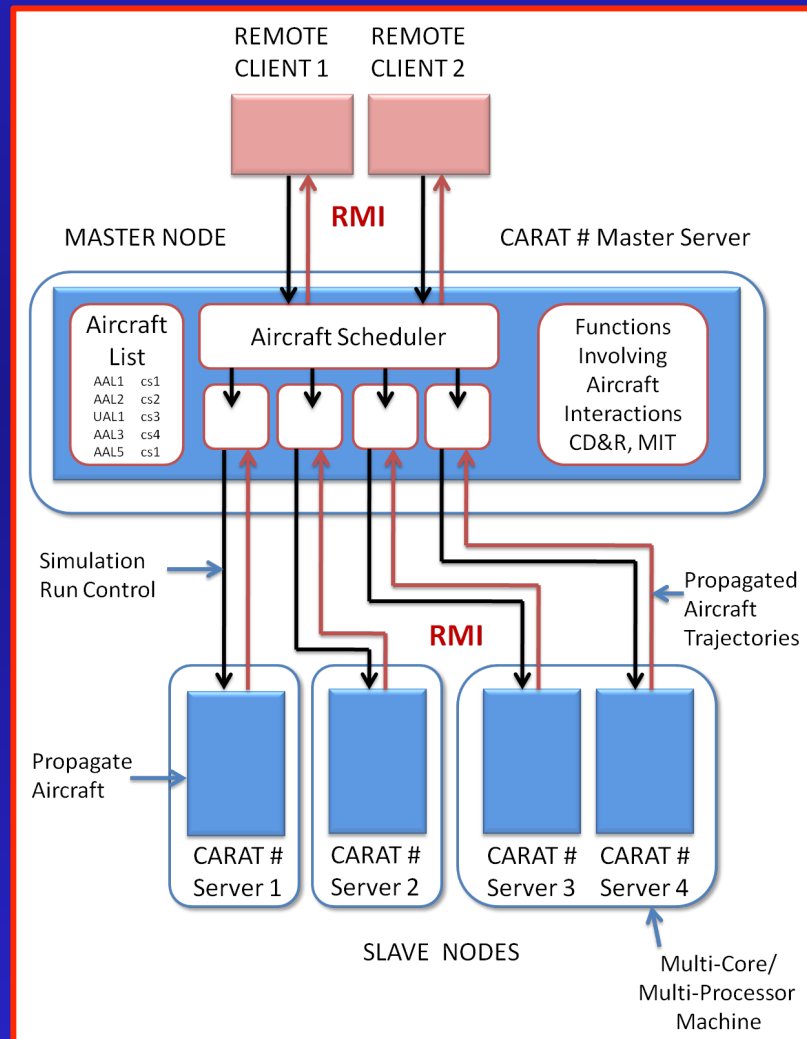
**3.1X faster**

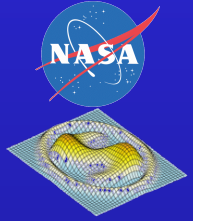


# Cluster Implementation

- Distribute computational load between multiple processors connected over a high-speed network
- FACET cluster implementation using Java RMI
- Max acceleration
  - Without data assembly at Master: **4.8x** (10 slaves)
  - With data assembly at Master: **2.4x** (12 slaves)
- Acceleration with software tuning included:

**5.3X faster**





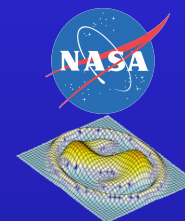
# Multi-Threaded FACET Implementation

---

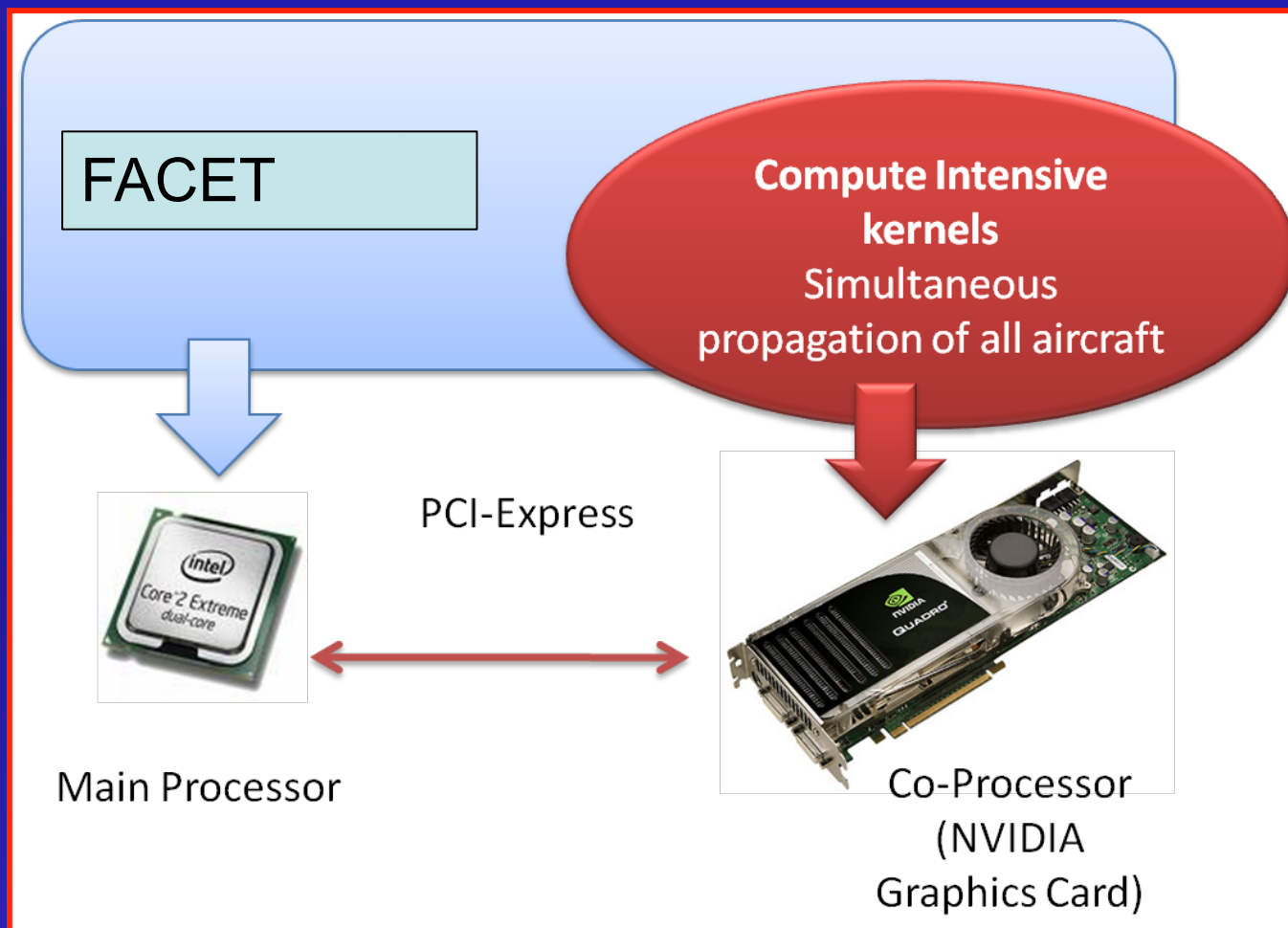
- Propagate aircraft for a single time step in parallel
- Divide aircraft list into  $N$  sub-lists for a  $N$ -core processor
- Run propagation loop over each sub-list concurrently in separate threads
- Implemented using POSIX threads

*8.0X faster*

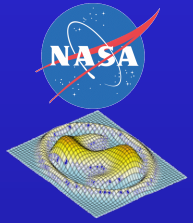
# **PARALLELIZING TRAFFIC FLOW SIMULATION FOR A GPU**



# Using the GPU as a Coprocessor

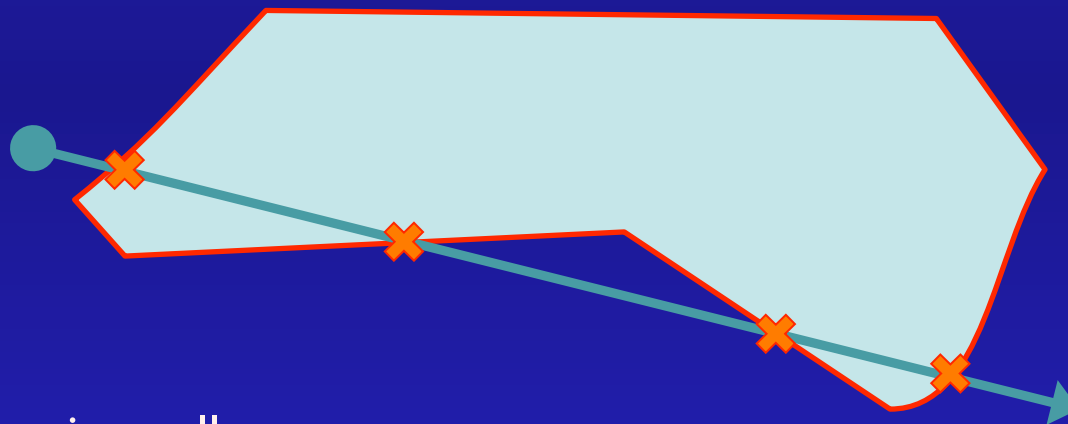






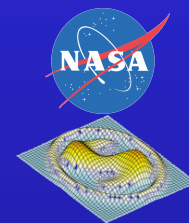
# GPU Implementation of `sim_inBoundary`

- Code Profile: functions that consumed most amount of runtime
- `sim_inBoundary`: Check whether a point is inside a polygon
- Ray Casting Algorithm



- Number of edges is small
- Data access time  $>$  time reduced by parallel computation over edges
- 24-hour simulation: 26 minutes

*10X slower*



# GPU as the Primary Processor

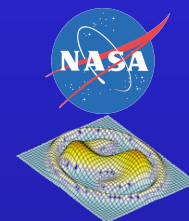
---

- Parallelization lesson learned
  - Want significant run time for single call
  - Need minimum amount of data transfer between host and GPU
- Propagation of every aircraft for single time step?

## Need for a new Trajectory Predictor

- FACET: Mixed C and Java Application
- FACET: Aircraft data contained in linked lists
- Develop CARPAT Trajectory Predictor completely in C

# **TRAJECTORY PREDICTOR**



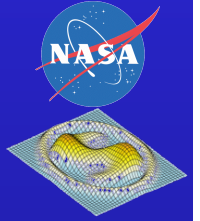
# Trajectory Predictor

---

- Flight data input: FACET track file

```
TRACK JBU222 A320 422700 853000 498 370 93 ZAU ZAU23  
FP_ROUTE LGB./PMM.J70.LVZ.LENDY5.JFK
```

- Aircraft performance data: BADA (Base of Aircraft Data)
- Airspace data such as airports, named waypoints, airways, sector boundary data, etc.



# Trajectory Predictor Model

- Calculation of great-circle heading between waypoints of the flight plan

$$\psi = \tan^{-1} \left( \frac{\sin(\tau_2 - \tau_1) \cos(\lambda_2)}{\sin(\lambda_2) * \cos(\lambda_1) - \sin(\lambda_1) * \cos(\lambda_2) * \cos(\tau_2 - \tau_1)} \right)$$

- Latitude propagation

$$\lambda_2 = \lambda_1 + \frac{V_G \cos(\psi)}{(R_E + h)} \Delta t$$

- Longitude propagation

$$\tau_2 = \tau_1 + \frac{V_G \sin(\psi)}{(R_E + h) \cos(\lambda_1)} \Delta t$$

- Ground speed

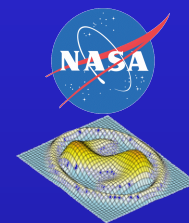
$$V_G$$

- Altitude propagation

$$h_2 = h_1 + \dot{h} \cdot \Delta t$$

- Climb/descent rate

$$\dot{h}$$

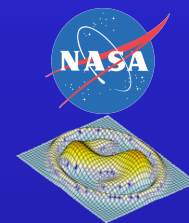


# Trajectory Output Data

---

## Flight data at every 30-second interval:

- Time since the start of simulation (seconds)
- Flight mode (Preflight, Climb, Cruise, Descent, Landed)
- Latitude (degrees): -90 to +90: North positive
- Longitude (degrees): -180 to +180: East positive
- Altitude (feet)
- True air speed (knots)
- Altitude rate (feet/second)
- Heading angle (degrees)
- Flight path angle (degrees)
- Sector Index



# Multicore Trajectory Predictor

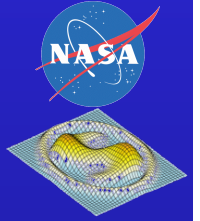
---

- Automatically detects number of cores on the computer
- Splits aircraft into lists to run in parallel on different cores
- Runtimes for 24-hour simulation

# Threads	Propagation Time (s) w/o thread affinity	Propagation Time (s) w/ thread affinity	Propagation Speed up over single thread	Trajectory output file writing (s)
1	65.47	66.31	-	36.96
2	55.88	37.57	1.76X	24.69
3	63.41	24.76	2.22X	18.81
4	65.66	24.76	2.68X	14.91

**PORTING CODE TO CUDA**

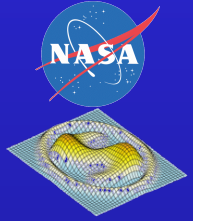




# Obstacles to GPU Implementation

---

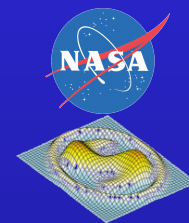
- Integer and Single-Precision Floating Point supported
- Dynamic memory allocation inside a structure is not allowed
- Host functions cannot be called from the device functions
- Recursive functions are not allowed
- Linked Lists cannot be used



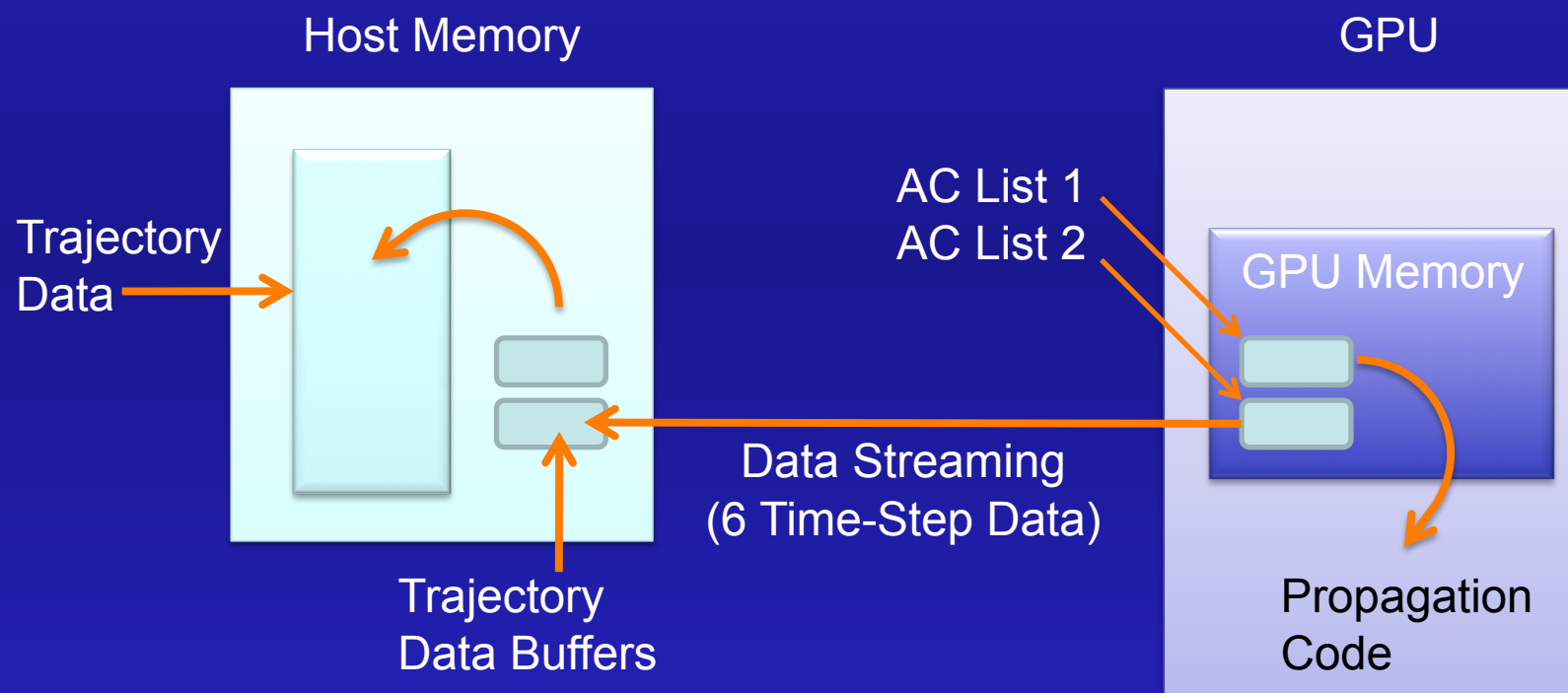
# Trajectory Predictor Data Storage

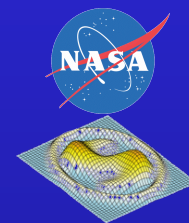
---

- All aircraft data is stored in global memory, because read/write access is needed
- Must minimize high-latency memory access
- Each aircraft has a large trajectory data structure
- Use streams to hide the latency



# Trajectory Data Streaming: GPU To Host

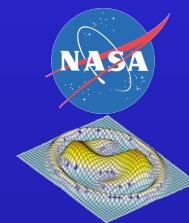




# Trajectory Data Streaming: GPU To Host

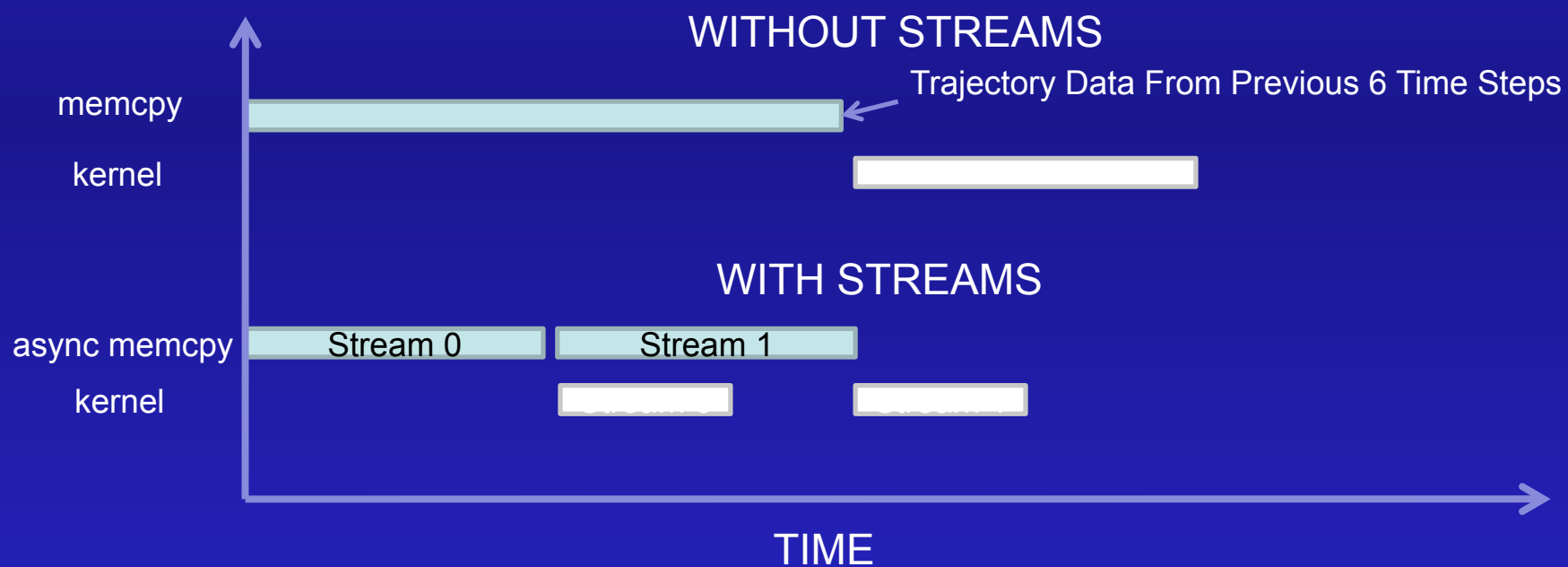
---

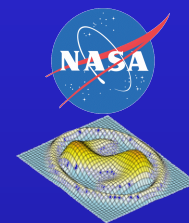
- Streams
  - Memory transfer and kernel execution of each list must happen sequentially, but memory transfer and kernel execution of separate lists can happen simultaneously
  - Requires asynchronous memory transfer
- Asynchronous Memory Transfer
  - Does not block CPU computations by default
  - CPU computations be blocked using `cudaStreamSynchronize()`
  - Host memory has to reside in the page-locked (pinned) memory



# Timeline Comparison

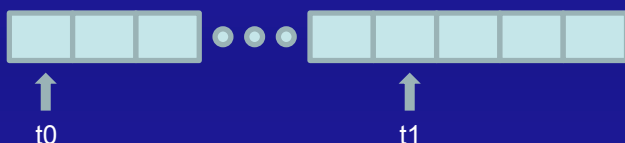
- After the trajectory data in GPU is full (6 time-step propagations):





# Memory Optimization

- Coalescing global memory access of aircraft data
  - Original implementation: array of structures
  - Modified Implementation: independent arrays for each field

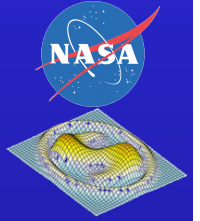


Original : prevents  
coalescing  
(size of structure is large)



Modified: leads to coalescing

- Use registers to avoid redundant memory transfers, whenever possible
  - Registers are limited, use caution when allocating them

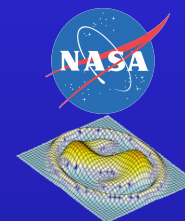


# Optimization of Parallelization

---

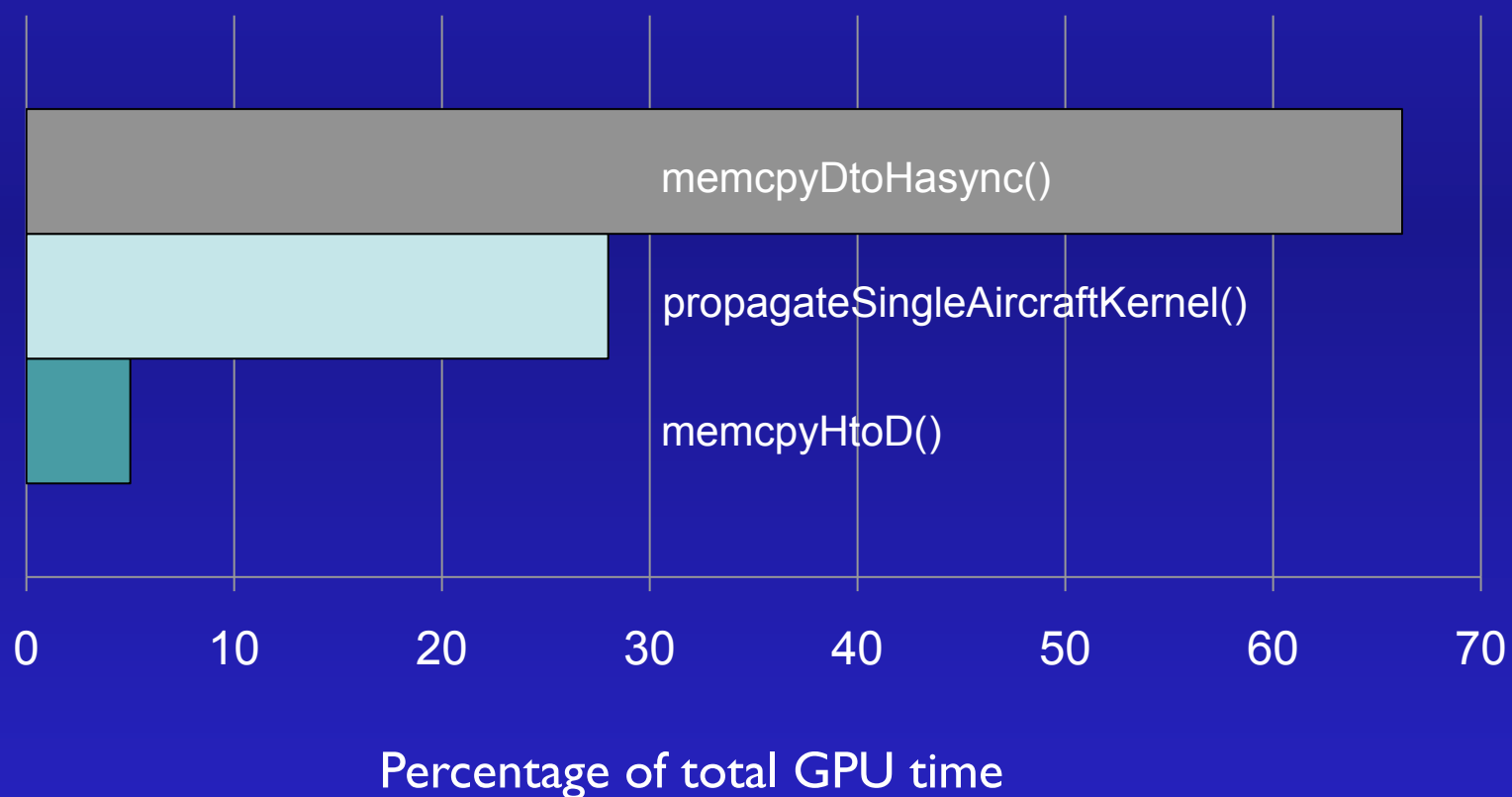
- Increase parallelization
  - Limiting the number of registers allows bigger blocks and improves thread parallelism
- Minimize branching within a warp
  - Reducing the if-else statements and while loops whenever possible
  - Performance lookup
    - Index is based on altitude, found using a 'while' loop
    - Instead, altitude table has 500-foot intervals, can be calculated:

$$index = \text{floor}\left(\frac{current\_altitude}{500}\right)$$



# CUDA Profile Summary

---





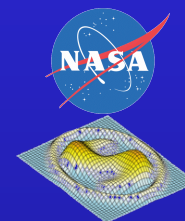
# **EXPERIMENTAL RESULTS**

# GPU Hardware

---

- GeForce 9800 GT (112 cores)
- TeslaC-1060 (240 cores)

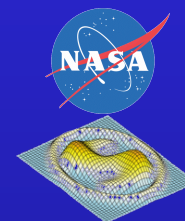




# GPU Comparison

	GeForce 9800GT	Tesla C 1060
# Cores	112	240
Processor clock	900 MHz	1300 MHz
Dedicated memory	0.5 GB	4 GB
Memory clock	900 MHz	800 MHz
Memory interface width	256 bit	512 bit
Memory bandwidth	57.6 GB/s	102 GB/s
<b>Simulation time</b>	<b>4.18 s</b>	<b>2.42 s</b>

*1.7X faster*

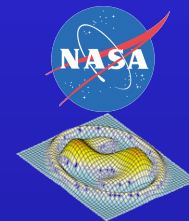


# GPU Implementation Results

Implementation	Runtime (s)	Speed-up w.r.t. FACET	Speed-up w.r.t. CPU	Speed-up w.r.t. Multi-threaded
Original FACET	586.9			
Trajectory Predictor (CPU)	46.6	12.6X		
Trajectory Predictor (Multi-threaded)	24.5	23.9X	1.9X	
Trajectory Predictor (GPU)	2.4	242.5X	19.2X	10.1X

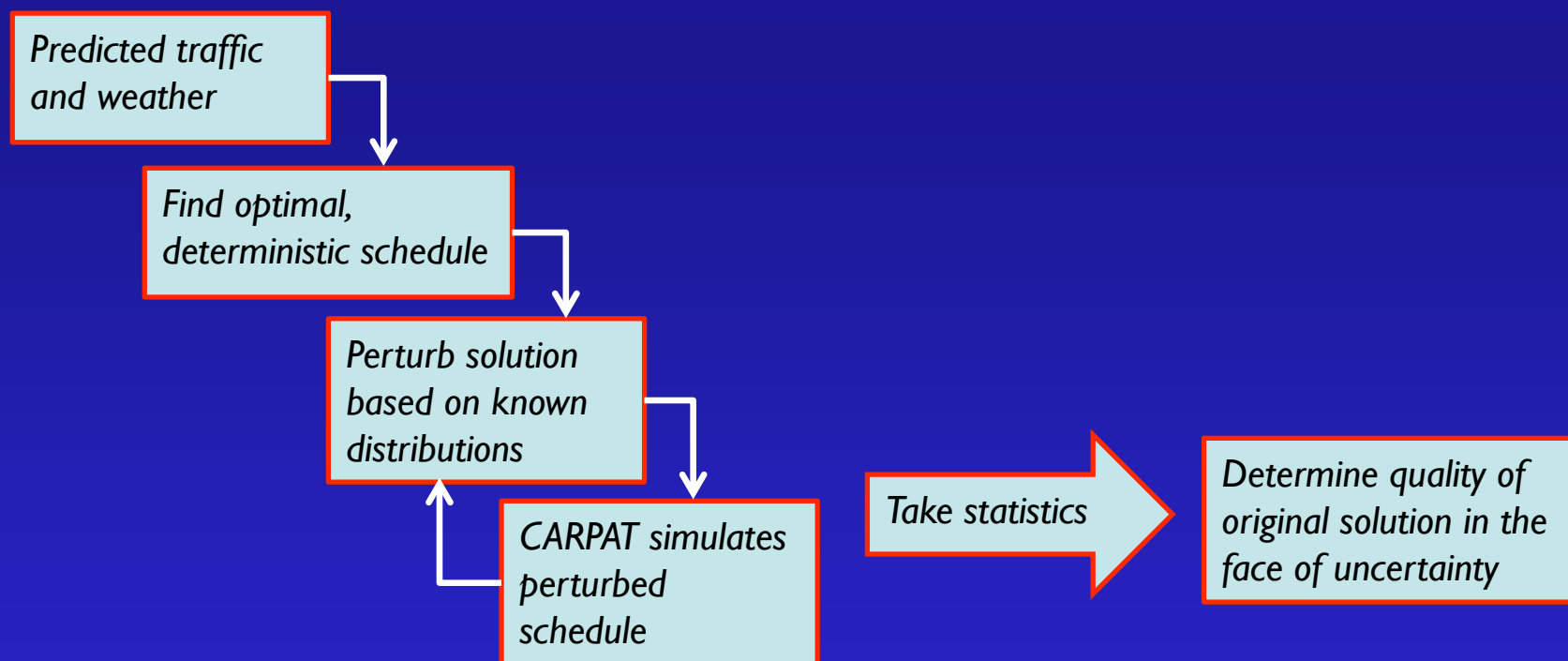
*243X faster*

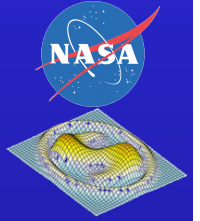
**POTENTIAL USES**



# Monte Carlo Simulations

- There is uncertainty in traffic flow management
- Many models for traffic flow management are deterministic
- Many uncertainties have been quantified in previous research





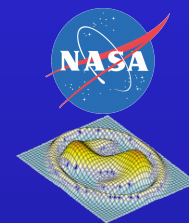
# Applications in Other Domains

---

- Traffic flow analysis of other transportation systems
  - Automobile
  - Train
  - Space
- Job-shop scheduling

**CONCLUDING REMARKS**





# Research Achievements

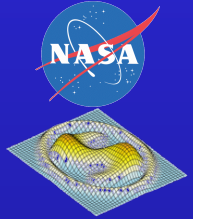
---

- Parallelized trajectory prediction code of a high-fidelity airspace simulator
- Ported parallel implementation to CUDA architecture
- Gained 240X speed-up over original implementation

*24-hour simulation with 35,000 aircraft  
completed under 2.5 seconds*

# Questions?

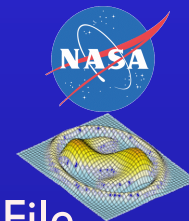
---



Joseph.L.Rios@nasa.gov  
*NASA Ames Research Center*

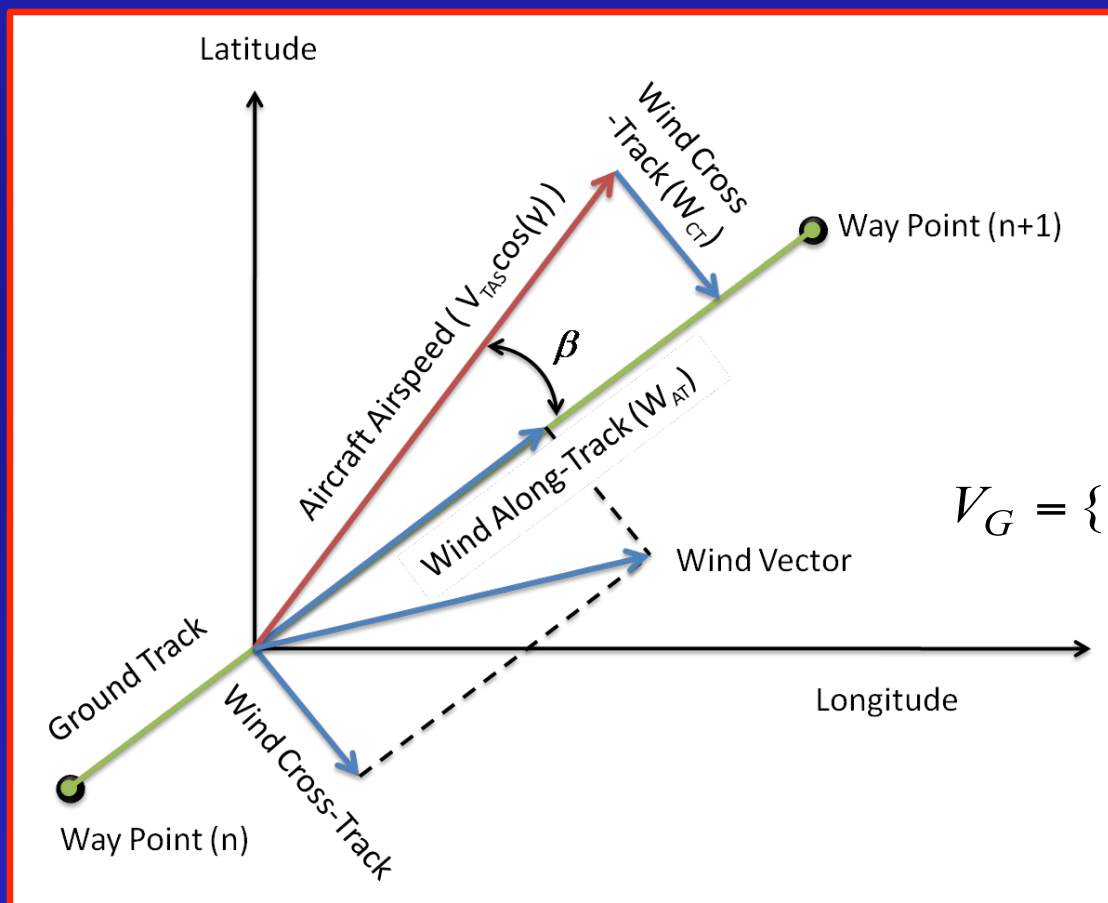
P. K. Menon, Monish D. Tandale & Sandy Wiraatmadja  
*Optimal Synthesis Inc.*

**BACKUP SLIDES**



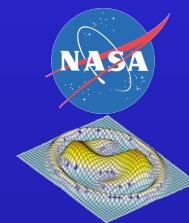
# Effect of Wind Along Track

- Reading True North and True East Wind Components from the RUC File



$$\beta = \sin^{-1} \left( \frac{W_{CT}}{V_{TAS} \cos(\gamma)} \right)$$

$$V_G = \{ V_{TAS} \cos(\gamma) \} \cos(\beta) + W_{AT}$$

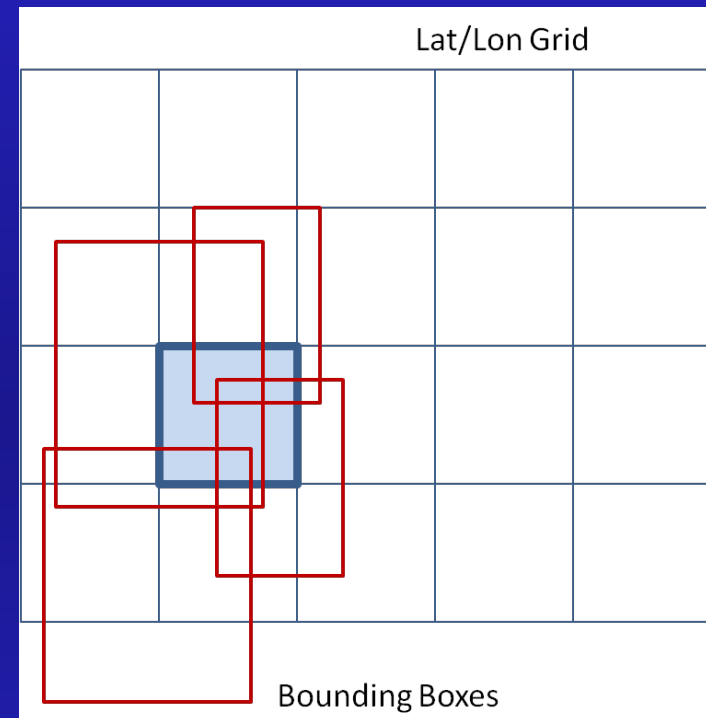
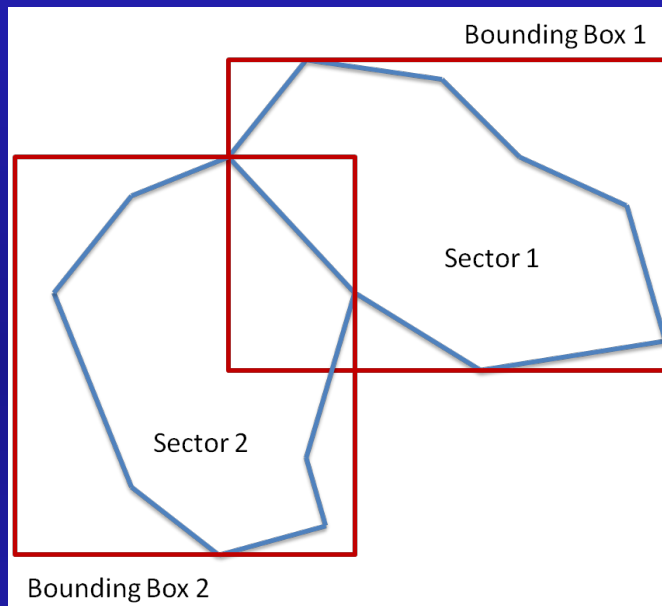
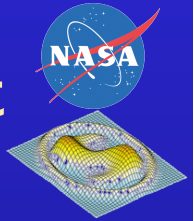


# Incorporating RUC Weather Data

---

- RUC: Rapid Update Cycle: NOAA/NCEP weather forecast
  - 13 km resolution, 50 Vertical Levels
- Wind Data: u & v components of the wind (Lambert Conformal Projection)
  - At a given 13km grid cell for a geopotential height
- Convert to True North and True East Wind Components at a given (Lat, Lon, Alt)
- Script to Download and Automatically Store RUC Data Files from the NOAA Website

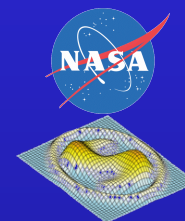
# Identifying the Current Sector for every Aircraft at every Time Step



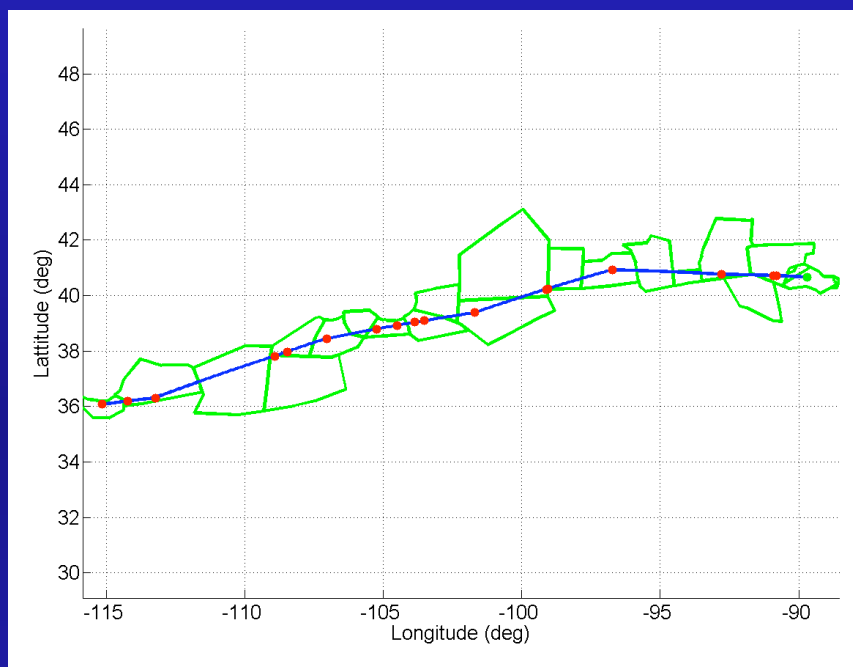
Check if Aircraft in Sector

- Check Within Bounding Box
- Ray Casting Algorithm

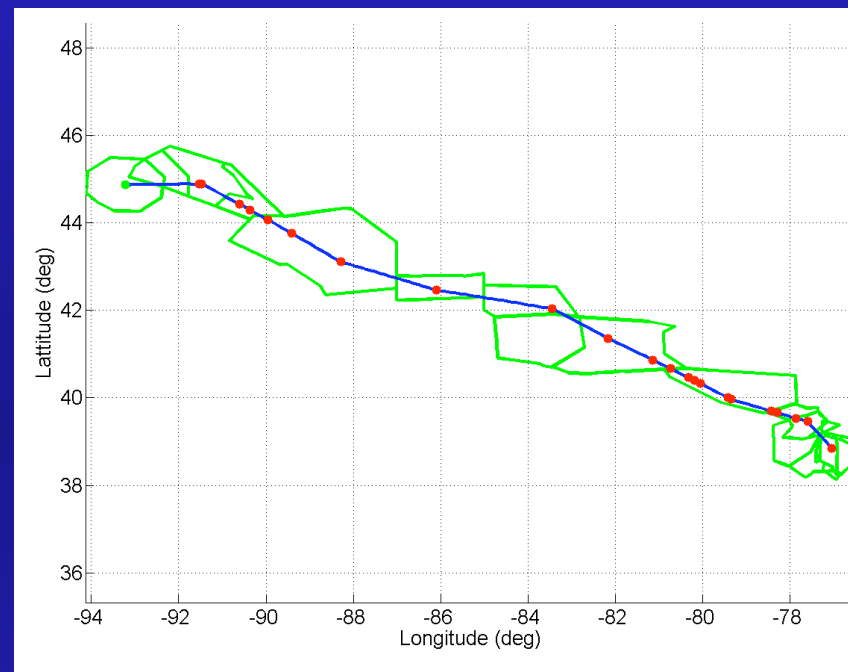
- 3D Hash Map:  $2^\circ$  Lat,  $2^\circ$  Lon, 1000 ft.
- For every cell, store all sectors whose bounding boxes overlap the cell



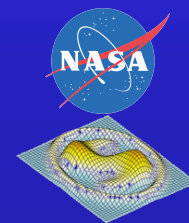
# Results of Sector Identification Code



Sequence of Sectors  
Traversed by Flight NWA1715



Sequence of Sectors  
Traversed by Flight AAY462



# Modeling Traffic Flow

---

- Aggregate models
  - Flights aggregated into flows
  - Optimal controls determine rates for entering/exiting flights
  - Solution needs disaggregation for implementation in reality
  - In general, computationally efficient
- Aircraft-level models
  - Controls considered for each flight
  - Solution is “implementation ready”
  - In general, computationally difficult
- Both approaches rely on airspace simulations